1. Record Nr. UNISALENTO991001323249707536 Autore Chern, Shiing-Shen **Titolo** Selected papers / Shiing-Shen Chern Pubbl/distr/stampa New York: Springer-Verlag, 1978 **ISBN** 0387903399 Descrizione fisica 476 p.; 24 cm. Classificazione AMS 53-XX Disciplina 514.7 Soggetti Collections of papers Differential geometry Global analysis Chern, Shiing-Shen Chern, Shiing-Shen Lingua di pubblicazione Inglese Formato Materiale a stampa Livello bibliografico Monografia

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Note generali

2. Record Nr. UNINA9911034942303321 Autore Öchsner Andreas **Titolo** A Simplified Approach to the Classical Laminate Theory of Composite Materials: Application of Bar and Beam Elements / / by Andreas Öchsner Cham:,: Springer Nature Switzerland:,: Imprint: Springer., 2025 Pubbl/distr/stampa **ISBN** 3-031-98052-2 Edizione [2nd ed. 2025.] Descrizione fisica 1 online resource (260 pages) Collana Advanced Structured Materials, , 1869-8441; ; 242 Disciplina 620.118 Soggetti Continuum mechanics Composite materials Mathematics - Data processing Continuum Mechanics Composites Computational Science and Engineering Lingua di pubblicazione Inglese **Formato** Materiale a stampa Livello bibliografico Monografia Introduction -- Bar Elements -- Euler-Bernoulli Beam Elements --Nota di contenuto Combination of Bar and Beam Elements -- Classical Laminate Theory for One-Dimensional Elements -- Example Problems -- Outlook to the Two-Dimensional Case. Sommario/riassunto This book provides a systematic introduction to composite materials, which are obtained by a layer-wise stacking of one-dimensional bar/beam elements. Each layer may have different mechanical properties, but each single layer is considered as isotropic. The major idea is to provide a simplified theory to easily understand the classical two-dimensional laminate theory for composites based on laminae with unidirectional fibers. In addition to the elastic behavior, failure is investigated based on the maximum stress, maximum strain, Tsai-Hill, and the Tsai-Wu criteria. Partial differential equations lay the foundation to mathematically describe the mechanical behavior of any

classical structural member known in engineering mechanics, including composite materials. The so-called classical laminate theory provides a simplified stress analysis, and a subsequent failure analysis, without

the solution of the system of coupled differential equations for the unknown displacements. The procedure provides the solution of a statically indeterminate system based on a generalized stress-strain relationship under consideration of the constitutive relationship and the definition of the so-called stress resultants. This laminate theory is typically provided for two-dimensional plane problems, where the basic structural element is a simple superposition of a classical plane elasticity element with a thin plate element under the consideration of an orthotropic constitutive law. This two-dimensional approach and the underlying advanced continuum mechanical modeling might be very challenging for some students, particularly at universities of applied sciences. Thus, a reduced approach, the so-called simplified classical laminate theory, has been developed. The idea is to use solely isotropic one-dimensional elements, i.e., a superposition of bar and beam elements, to introduce the major calculation steps of the classical laminate theory. Understanding this simplified theory is much easier and the final step it to highlight the differences when moving to the general two-dimensional case.