

- | | |
|-------------------------|---|
| 1. Record Nr. | UNIORUON00103690 |
| Titolo | Index sinicus : A catalogue of article relating to China in periodicals and other collective publications 1920-1955 / [a cura di] John Lust |
| Pubbl/distr/stampa | Cambridge, : W. Heffer & S ons Ltd., 1964 |
| Descrizione fisica | XXX, 663 p. ; 24 cm |
| Classificazione | CIN GEN B I |
| Lingua di pubblicazione | Inglese |
| Formato | Materiale a stampa |
| Livello bibliografico | Monografia |
-
- | | |
|-------------------------|--|
| 2. Record Nr. | UNINA9911006598703321 |
| Autore | Hughes Peter C |
| Titolo | Spacecraft Attitude Dynamics |
| Pubbl/distr/stampa | Newburyport, : Dover Publications, 2012 |
| ISBN | 0-486-14013-X
1-62198-601-2 |
| Edizione | [1st ed.] |
| Descrizione fisica | 1 online resource (1137 p.) |
| Collana | Dover Books on Aeronautical Engineering |
| Disciplina | 629.4742 |
| Soggetti | Space vehicles - Attitude control systems |
| Lingua di pubblicazione | Inglese |
| Formato | Materiale a stampa |
| Livello bibliografico | Monografia |
| Note generali | Description based upon print version of record. |
| Nota di contenuto | 3.7 Dynamics of a System of Rigid Bodies3.8 Problems; Chapter 4 Attitude Dynamics of a Rigid Body; 4.1 Basic Motion Equations; 4.2 Torque-Free Motion; R Inertially Axisymmetrical; 4.3 Torque-Free Motion; R Tri-inertial; 4.4 Stability of Motion for R; 4.5 Motion of a Rigid Body Under Torque; 4.6 Problems; Chapter 5 Effect of Internal Energy Dissipation on the Directional Stability of Spinning Bodies; 5.1 Quasi-Rigid Body with an Energy Sink, R; 5.2 Rigid Body with a Point Mass Damper, R + R; 5.3 Problems; Chapter 6 Directional Stability of |

Multispin Vehicles; 6.1 The R + W Gyrostat
 6.2 Gyrostat with Nonspinning Carrier 6.3 The Zero Momentum
 Gyrostat; 6.4 The General Case; 6.5 System of Coaxial Wheels; 6.6
 Problems; Chapter 7 Effect of Internal Energy Dissipation on the
 Directional Stability of Gyrostats; 7.1 Energy Sink Analyses; 7.2
 Gyrostats with Discrete Dampers; 7.3 Problems; Chapter 8 Spacecraft
 Torques; 8.1 Gravitational Torque; 8.2 Aerodynamic Torque; 8.3
 Radiation Torques; 8.4 Other Environmental Torques; 8.5
 Nonenvironmental Torques; 8.6 Closing Remarks; 8.7 Problems;
 Chapter 9 Gravitational Stabilization; 9.1 Context
 9.2 Equilibria for a Rigid Body in a Circular Orbit 9.3 Design of
 Gravitationally Stabilized Satellites; 9.4 Flight Experience; 9.5 Problems;
 Chapter 10 Spin Stabilization in Orbit; 10.1 Spinning Rigid Body in
 Orbit; 10.2 Design of Spin-Stabilized Satellites; 10.3 Long-Term Effects
 of Environmental Torques, and Flight Data; 10.4 Problems; Chapter 11
 Dual-Stabilization in Orbit: Gyrostats and Bias Momentum Satellites;
 11.1 The Gyrostat in Orbit; 11.2 Gyrostats with External Rotors; 11.3
 Bias Momentum Satellites; 11.4 Problems; Appendix A Elements of
 Stability Theory; A.1 Stability Definitions
 A.2 Stability of the Origin A.3 The Linear Approximation; A.4 Nonlinear
 Inferences from Infinitesimal Stability Properties; A.5 Liapunov's
 Method; A.6 Stability of Linear Stationary Mechanical Systems; A.7
 Stability Ideas Specialized to Attitude Dynamics; Appendix B Vectrices;
 B.1 Remarks on Terminology; B.2 Vectrices; B.3 Several Reference
 Frames; B.4 Kinematics of Vectrices; B.5 Derivative with Respect to a
 Vector; Appendix C List of Symbols; C.1 Lowercase Symbols; C.2
 Uppercase Symbols; C.3 Lowercase Greek Symbols; C.4 Uppercase
 Greek Symbols; C.5 Other Notational Conventions; References
 Errata

Sommario/riassunto

Pointing a satellite in the right direction requires an extremely complex system - one that describes the satellite's orientation and at the same time predicts and either uses or neutralizes external influences. From its roots in classical mechanics and reliance on stability theory to the evolution of practical stabilization ideas, Spacecraft Attitude Dynamics offers comprehensive coverage of environmental torques encountered in space; energy dissipation and its effects on the attitude stability of spinning bodies; motion equation for four archetypical systems derived and used repeatedly throu